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EPTT-2022-XXXX REVISED AND EXTENDED RESULTS ON A LAMINAR PERTURBED FLOW TOWARDS CALIBRATION OF A KINETIC ENERGY – HELICITY TURBULENCE MODEL

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Abstract. Fluctuating velocities and pressures at selected points of an oscillated laminar flow are recoded and processed to provide the statistical correlations of the transport equations for kinetic energy, helicity and enstrophy. The primitive variables solver uses UNIFAES discretization for the advective and viscous terms. Such scheme is generalized by making explicit the space derivatives employed in the solver, allowing its use in the calculation of the various correlations, in parallel with second order and forth order central differencing schemes.

Keywords: Generalized UNIFAES, kinetic energy, helicity, enstrophy.

1. INTRODUCTION

The construction of a turbulence model based on fluctuating kinetic energy and helicity was proposed by Figueiredo (2018b). Helicity is the scalar product of velocity and vorticity. Such concept was formulated about 1960, with applications in meteorology and magneto-hydrodynamics (Moffatt and Tsinober, 1992). Fluctuations in helicity have been associated to turbulence cascade and vortex stretching in physics literature (Holm, 2007; Dallas and Tobias, 2016; Yan et al., 2020). However, helicity is absent from the literature on statistical models of turbulence for engineering.

The primitive variables incompressible Navier-Stokes solver applies to three-dimensional flows, using UNIFAES scheme for the advective and viscous transport terms, semi-staggered mesh, Poisson equation for pressure with momentum interpolation, and forth order Runge-Kutta time-wise integration. Instantaneous velocity and pressure around selected nodes are recorded to be processed in the program for the statistical correlations of the transport equations for kinetic energy, helicity and enstrophy. Testing of the statistics program used an oscillated laminar flow, which can be analyzed with modest refinement levels allowed by a personal computer.

Initial results concerning this proposal (Figueiredo, 2020) are revised, an important error on the helicity transport equation is corrected, and, particularly, coherence between the numerical methods of the Navier-Stokes solver and those of the statistical correlations is improved through a generalization of UNIFAES which turned explicit its first and second derivatives, so that the correlations can be calculated with the same interpolating curve of the solver.

2. GENERALIZED UNIFAES SCHEME

UNIFAES computes the combined advective and diffusive terms without requiring its first and second derivatives, which are made explicit here. This work is restricted to regularly spaced Cartesian meshes, although UNIFAES admits irregularly spaced meshes without loss of second order convergence, with adequate refinement route (Llagostera and Figueiredo, 2000a, 2000b).

UNIFAES applies to the divergence form advective term and diffusive term of the transport equation of a transported variable Φ on a velocity field U_i :

$$\frac{\partial \Phi}{\partial t} + \frac{\partial u_i \Phi}{\partial x_i} - \nu \frac{\partial^2 \Phi}{\partial x_i \partial x_i} = S \tag{1}$$

Here, ν is the diffusivity of the transported variable and S is a possible source term. Summation over the three coordinate directions is assumed for each repeated index in a term.

The Finite Volume analogue of the net advective and diffusive flux $\partial J_1 / \partial x_1$ is:

$$\frac{\partial J_1}{\partial x_1} \cong \frac{J_e - J_w}{\Delta x_1} = \frac{1}{\Delta x_1 \Delta x_2 \Delta x_3} \int_{x3_d}^{x3_u} \int_{x2_s}^{x2_n} \int_{x1_w}^{x1_e} \left(\frac{\partial U \Phi}{\partial x_1} - v \frac{\partial^2 \Phi}{\partial x_1^2} \right) dx_1 dx_2 dx_3 \cong \frac{1}{\Delta x_1} \left[\mathcal{U}_e \Phi_e - v \frac{\partial \Phi}{\partial x_{1_e}} - \mathcal{U}_w \Phi_w + v \frac{\partial \Phi}{\partial x_{1_w}} \right]$$
(2)

Figueiredo, J. R. Revised and Extended Results on a Laminar Perturbed Flow Towards Calibration of a Kinetic Energy – Helicity Turbulence Model

Division by cell volume is used above to restore the physical dimension of the original equation, as convenient for expressing derivatives. Exponential-type schemes employ interpolating curves obtained as exact solutions of the one-dimensional linear equation:

$$\mathcal{U}\frac{d\Phi}{dx} - \nu\frac{d^2\Phi}{dx^2} = K \tag{3}$$

Equation (3) approximates transport equation (1) considering the velocity component \mathcal{U} in direction x to be locally constant, as well as term K, which represents all other terms of the equation (1), namely cross flow transport, source and transient terms. The resulting interpolation curve is:

$$\Phi = C_1 + C_2 exp\left(\frac{ux}{v}\right) + \frac{\kappa}{u}x\tag{4}$$

The Finite Volume Exponential-type schemes employ the interpolating curve (4) to determine the flux through cell face *e*, for instance, using reference properties at such face. Curve (4) is fit to nodes P e E, with origin at node P, yielding:

$$C_2 = \frac{\Phi_{\rm E} - \Phi_{\rm P} - \frac{K_{\rm e}}{u_{\rm e}} \Delta x}{\exp\left(\frac{u_{\rm e}\Delta x}{v}\right) - 1} \tag{5}$$

$$C_1 = \Phi_P - C_2 \tag{6}$$

Deriving $\Phi(x)$ from Equation (3) :

$$\frac{d\Phi}{dx} = C_2 \frac{u}{v} exp\left(\frac{ux}{v}\right) + \frac{\kappa}{u} \tag{7}$$

Profile (4) and its derivative (7) are then employed to compute the advective – diffusive flux. At cell face *e* it results, in terms of the cell face Peclet number $p_e = U_e \Delta x_1 / v$:

$$J_e = \mathcal{U}_e \Phi_e - \nu \frac{\partial \Phi}{\partial x_e} = \mathcal{U}_e \mathcal{C}_1 + K \left(x_e - \frac{\nu}{\mathcal{U}_e} \right) = (\Phi_P - \Phi_E) \nu \frac{\pi(p_e)}{\Delta x_1} + K_e \left[\frac{\pi(p_{e/w}) - 1}{p_{e/w}} + \frac{1}{2} \right] + \mathcal{U}_e \Phi_P \tag{8}$$

Proceeding analogously for the west side and substituting into (2), the combined advective-diffusive net flux through east and west faces is:

$$\frac{J_e - J_W}{\Delta x_1} \cong \frac{1}{\Delta x_1} \left[\nu \frac{\pi(p_e)}{\Delta x_1} (\Phi_P - \Phi_E) + \nu \frac{\pi(p_w)}{\Delta x_1} (\Phi_P - \Phi_W) + \Psi + (\mathcal{U}_e - \mathcal{U}_w) \Phi_P \right]$$
(9)

$$\pi(p_{e/w}) = \frac{p_{e/w}}{\exp(p_{e/w}) - 1} \tag{10}$$

$$\Psi = K_e \chi(p_e) - K_w \chi(p_w) \tag{11}$$

$$\chi(p_{e/w}) = \frac{\pi(p_{e/w}) - 1}{p_{e/w}} + \frac{1}{2}$$
(12)

Term $(\mathcal{U}_e - \mathcal{U}_w)\Phi_P/\Delta x_1$ and its analogues for other directions vanish due to continuity.

Terms *K* are determined in UNIFAES by Allen and Southwell's (1955) Exponential scheme. Unknowns C_1 , C_2 and K_P/U_P of curve (4) are fit to the nodes W, P and E, using properties of node P, forming a system that can be solved for K_P in terms of cell Peclet number $p_P = U_P \Delta x/v$:

$$K_P = (\Phi_P - \Phi_E) \frac{\pi(p_P)}{\Delta x^2} + (\Phi_P - \Phi_W) \frac{\pi(-p_P)}{\Delta x^2}$$
(13)

$$\pi(p_P) = \frac{p_P}{exp(p_P) - 1} \tag{14}$$

UNIFAES determines K_e by interpolating between the estimates of K_P , Eq. (13), and analogous for K_E . If node E is located at a wall, K_e is obtained by extrapolation from K_P and K_W .

The Finite Volume analogues of the first and second derivatives in direction x_1 are obtained by integration on the cell volume using the divergence theorem:

$$\frac{1}{\Delta x_1} \int_{x_1 w}^{x_1 e} \frac{\partial \Phi}{\partial x_1} dx_1 = \frac{\Phi_e - \Phi_w}{\Delta x_1}$$
(15)

$$\frac{1}{\Delta x_1} \int_{x_1 w}^{x_1 e} \frac{\partial^2 \Phi}{\partial x_1^2} dx_1 = \left(\frac{\partial \Phi}{\partial x_1 e} - \frac{\partial \Phi}{\partial x_1 w}\right) \frac{1}{\Delta x_1}$$
(16)

From the interpolating curve (4), using constants (5) and (6), one obtains for the cell face e:

$$\Phi_{\rm e} = \Phi_{\rm P} + \frac{\Phi_{\rm E} - \Phi_{\rm P} - \frac{K_e}{u_e \Delta x}}{exp(p_{\rm e}) - 1} \left[exp(p_{\rm e}/2) - 1 \right] + \frac{K_e}{u_e} \frac{\Delta x}{2}$$
(17)

Proceeding analogously for the west side, one gets UNIFAES first derivative:

$$\frac{\Phi_e - \Phi_w}{\Delta x_1} = \frac{(\Phi_E - \Phi_P)\Theta(p_e) - (\Phi_W - \Phi_P)\Theta(-p_w)}{\Delta x_1} + \frac{K_e}{v}\Delta x.\,\Omega(p_e) - \frac{K_w}{v}\Delta x.\,\Omega(-p_w)$$
(18)

$$\Theta(p) = \frac{exp(p/2) - 1}{exp(p) - 1} = \frac{1}{exp(p/2) + 1}$$
(19)

$$\Omega(p) = \frac{\frac{1}{2} - \Theta(p)}{p} = \frac{exp(p/2) - 1}{2p(exp(p/2) + 1)} = \frac{\Theta(p)}{4\pi(p/2)}$$
(20)

Using Eq. (7) with constants (5) and (6), one obtains for the cell face *e*:

$$\frac{\partial \Phi}{\partial x_e} = \frac{\Phi_{\rm E} - \Phi_{\rm P} + \frac{K_e}{u_e \Delta x}}{exp(p_e) - 1} \cdot \frac{p_e}{\Delta x} \cdot exp\left(\frac{p_e}{2}\right) + \frac{K_e}{u_e}$$
(21)

By proceeding analogously for face *w* one obtains UNIFAES second derivative:

$$\left(\frac{\partial \Phi}{\partial x_{e}} - \frac{\partial \Phi}{\partial x_{w}}\right)\frac{1}{\Delta x} = \frac{\Phi_{E} - \Phi_{P}}{\Delta x^{2}}\lambda(p_{e}) + \frac{\Phi_{W} - \Phi_{P}}{\Delta x^{2}}\lambda(-p_{w}) + \frac{K_{e}}{v}\cdot\Upsilon(p_{e}) + \frac{K_{w}}{v}\cdot\Upsilon(-p_{w})$$
(22)

$$\lambda(p) = \frac{p.exp\left(\frac{p}{2}\right)}{exp(p)-1} = \pi(p)exp\left(\frac{p}{2}\right)$$
(23)

$$\Upsilon(p) = \frac{1 - \pi(p) exp\left(\frac{p}{2}\right)}{p} \tag{24}$$

Paradoxically, the generalized UNIFAES derivatives are non-linear difference operators representing linear differential operators. Determining the UNIFAES derivative of the fluctuating component requires computing derivatives of instantaneous velocities at instantaneous conditions and derivatives of mean velocities at mean conditions, as indicated in the right side of eq. (25):

$$\left|\frac{\widetilde{\partial u_i}}{\partial x_j}\right| = \left|\frac{\partial(\widetilde{u_i+u_i})}{\partial x_j}\right|_{U_j+u_j} - \left|\frac{\partial\widetilde{u_i}}{\partial x_j}\right|_{U_j}$$
(25)

Above procedure differs from applying UNIFAES instantaneous derivation directly to the fluctuating velocity components, as expressed in Eq. (26):

$$\left|\frac{\partial \widetilde{u}_{l}}{\partial x_{j}}\right|_{U_{j}+u_{j}} = \left|\frac{\partial (\widetilde{u_{t}+u_{l}})}{\partial x_{j}}\right|_{U_{j}+u_{j}} - \left|\frac{\partial \widetilde{u}_{l}}{\partial x_{j}}\right|_{U_{j}+u_{j}}$$
(26)

In cases with steady mean conditions, linear approximation of UNIFAES differences can be defined by using only the local mean velocity in the cell Peclet number, as follows:

$$\left|\frac{\partial u_{\iota}}{\partial x_{j}}\right|_{U_{j}} = \left|\frac{\partial (\overline{u_{\iota}+u_{\iota}})}{\partial x_{j}}\right|_{U_{j}} - \left|\frac{\partial U_{\iota}}{\partial x_{j}}\right|_{U_{j}}$$
(27)

3. TRANSPORT EQUATIONS

The transport equation for fluctuating kinetic energy can be written according to the advective – diffusive structure (28), or by expressing the viscous terms with the shear rate, as (29):

$$\frac{\partial U_j \overline{u_l u_l/2}}{\partial x_j} - \nu \frac{\partial^2 \overline{u_l u_l/2}}{\partial x_j \partial x_j} = -\overline{u_l u_j} S_{ij} - \frac{\partial \overline{u_j p}}{\partial x_j} - \frac{\partial \overline{u_j u_l u_l/2}}{\partial x_j} - \nu \frac{\overline{\partial u_l}}{\partial x_j} \frac{\partial u_l}{\partial x_j}$$
(28)

$$\frac{\partial U_j \overline{u_l u_l/2}}{\partial x_j} - 2\nu \frac{\partial \overline{u_l s_{lj}}}{\partial x_j} = -\overline{u_l u_j} S_{ij} - \frac{\partial \overline{u_j p}}{\partial x_j} - \frac{\partial \overline{u_j u_l u_l/2}}{\partial x_j} - 2\nu \overline{s_{lj} s_{lj}}$$
(29)

The transport equation of the "squared vorticity", or enstrophy, $\overline{w_l w_l/2}$, is (Tennekes and Lumley, 1972):

$$\frac{\partial U_j \overline{w_l w_l/2}}{\partial x_j} - \frac{1}{Re} \frac{\partial^2 \overline{w_l w_l/2}}{\partial x_j \partial x_j} = \overline{w_l S_{lj}} W_j - \overline{w_l u_j} \frac{\partial W_i}{\partial x_j} + \overline{w_l w_j} S_{ij} - \frac{\partial \overline{u_j w_l w_l/2}}{\partial x_j} - \overline{w_l w_j S_{lj}} - \frac{1}{Re} \frac{\partial \overline{w_l} \partial w_l}{\partial x_j}$$
(30)

The transport equation for the velocity-vorticity tensor is deduced by Figueiredo (2018b, eq. 19):

$$\frac{\partial \overline{u_i w_k} U_j}{\partial x_j} - \nu \frac{\partial^2 \overline{u_i w_k}}{\partial x_j \partial x_j} = \overline{u_i w_j} S_{kj} - \overline{u_j w_k} S_{ij} + \left(\overline{u_i s_{kj}} - \frac{1}{2} \varepsilon_{ijl} \overline{u_l w_k}\right) W_j - \frac{\partial \overline{u_i w_k}}{\partial x_j} - \overline{w_k} \frac{\partial p}{\partial x_i} + \overline{u_i w_j s_{kj}} - \frac{\partial \overline{u_i w_k u_j}}{\partial x_j} - 2\nu \frac{\partial \overline{u_i} \frac{\partial w_k}{\partial x_j}}{\partial x_j}$$
(31)

The transport equation for helicity is obtained by contracting indexes i, k into i, i, yielding:

$$\frac{\partial \overline{u_i w_l} U_j}{\partial x_j} - \nu \frac{\partial^2 \overline{u_l w_l}}{\partial x_j \partial x_j} = \frac{\partial \overline{u_i u_j}}{\partial x_j} W_i - \overline{u_l u_j} \frac{\partial W_i}{\partial x_j} - \overline{w_l \frac{\partial p}{\partial x_l}} + \overline{u_l w_j S_{lj}} - \frac{\partial \overline{u_l w_l u_j}}{\partial x_j} - 2\nu \frac{\partial \overline{u_l} \frac{\partial w_l}{\partial x_j}}{\partial x_j \partial x_j}$$
(32)

Above expression corrects Eq. (20) of Figueiredo (2018b) with respect to the term proportional to mean vorticity W_i . The error was repeated at Eq. (22) of Figueiredo (2020) and caused slow convergence of the helicity transport equation in the results presented there. Derivation of this term is presented now. After contraction, the factor multiplying the mean vorticity W_i in Eq. (19) results:

$$\overline{u_l s_{lj}} - \frac{1}{2} \varepsilon_{ijl} \overline{u_l w_l} = \overline{u_l s_{lj}} - \frac{1}{2} \varepsilon_{lji} \overline{u_l w_l} = \overline{u_l s_{lj}} + \overline{u_l r_{jl}} = \overline{u_l s_{lj}} - \overline{u_l r_{lj}} = u_l \frac{\partial u_j}{\partial x_l} = \frac{\partial u_l u_j}{\partial x_l}$$
(33)

Noticeably, the production terms of helicity equation are related to the mean vorticity and its gradient, while the production term of the kinetic energy equation is related to the mean shear, so that both components of the mean velocity gradient tensor are complementarily considered.

4. RESULTS

Computations refer to the Couette flow sketched at Fig. 1, with normalized dimensions $4x_1x_1$, having lower and upper impermeable adherent walls, with normalized velocity $U_1 = 1$ at $x_2 = 1$, periodic conditions in span-wise direction x_3 , Reynolds number 600. Inlet and initial conditions are $U_i = U_1(x_2)e_1 = x_2e_1$, where e_i is versor in direction *i*. Cubic cells are adopted.



Figure 1 – Domain composed by perturbed field, measurements region and hyper-viscous field.

In the region $0 < x_1 \le 1$, an artificial oscillatory field f induces periodic rotation around x_3 axis:

$$f = M.\sin(2\pi\mu t).\beta.\gamma \tag{34}$$

$$\beta = exp\left\{-\left[\frac{\left(x_{1}-x_{1}^{O}\right)^{2}}{\sigma_{1}^{2}} + \frac{\left(x_{2}-x_{2}^{O}\right)^{2}}{\sigma_{2}^{2}} + \frac{\left(x_{3}-x_{3}^{O}\right)^{2}}{\sigma_{3}^{2}}\right]\right\}$$
(35)

$$\gamma = [-(x_2 - x_2^0)e_1 + (x_1 - x_1^0)e_2]$$
(36)

where $(x_1^0, x_2^0, x_3^0) = (0.5, 0.5, 0.5), M = 2, \sigma_1 = \sigma_2 = 0.2, \sigma_3 = 0.4$ and $\mu = 0.5$. In the last part of domain, $3 \le x_1 < 4$, an hyper viscous fluid reduces the Reynolds number to 10 in order to enforce a steady flow compatible with homogeneous Newman conditions at outlet.

Table 1 presents the values of kinetic energy, helicity and enstrophy in 18 nodes in planes $x_1=1.5$ and $x_1=2.5$, for Reynolds number 600 with mesh 240x60x60. The values of helicity and enstrophy depend on the discretization; generally differences are around or below 1%, except at nodes 1 and 3, with differences about 30%. Both forms of UNIFAES present almost coincident results, which are closer to the forth order central differencing than to the second order one.

Table 1. Fluctuations statistics at selected nodes for Re=600 for mesh 240x60x60.

Desition			<i>x</i> ₁ = 1.5		$x_1 = 2.5$			
	Position	$x_3 = 0.25$	<i>x</i> ₃ = 0.5	$x_3 = 0.75$	$x_3 = 0.25$	$x_3 = 0.5$	$x_3 = 0.75$	
	Node	7	8	9	16	17	18	
	Kinetic	x10 ⁻⁴	10-4	10-4	10-4	10-4	10-4	
<i>x</i> ₂	Energy	2.349	3.846	2.349	1.216	1.108	1.216	
= 75	Helicity	10-3	10-11	10-3	10-3	10-10	10-3	
.75	CD 2 nd	-1.503	3.515	1.503	-1.566	1.057	1.566	
	CD 4 th	-1.493	3.561	1.493	-1.562	1.060	1.562	
	Uni NL	-1.496	3.557	1.496	-1.563	1.060	1.563	
	Uni L	-1.496	3.558	1.496	-1.563	1.060	1.563	
	Enstrophy	10-2	10-2	10-2	10-2	10-3	10-2	
	CD 2 nd	1.038	1.024	1.038	1.533	6.280	1.533	
	CD 4 th	1.047	1.047	1.047	1.534	6.286	1.534	
	Uni NL	1.046	1.042	1.046	1.535	6.289	1.535	
	Uni L	1.046	1.042	1.046	1.535	6.290	1.535	
	Node	4	5	6	13	14	15	
	Kinetic	10-3	10-3	10-3	10-4	10-4	10-4	
<i>x</i> ₂	Energy	1.127	1.095	1.127	2.756	1.291	2.756	
=	Helicity	10-2	10-10	10-2	10-3	10-10	10-3	
.50	CD 2 nd	-2.659	5.703	2.659	-7.836	5.332	7.836	
	CD 4 th	-2.671	7.536	2.671	-7.916	5.350	7.916	
	Uni NL	-2.669	7.535	2.669	-7.897	5.350	7.897	
	Uni L	-2.669	7.536	2.669	-7.898	5.351	7.898	
	Enstrophy	10-1	10-1	10-1	10-2	10-3	10-2	
	CD 2 nd	1.999	1.271	1.999	6.101	3.776	6.101	
	CD 4 th	2.022	1.285	2.022	6.240	3.807	6.240	
	Uni NL	2.018	1.283	2.018	6.208	3.804	6.208	
	Uni L	2.018	1.283	2.018	6.208	3.805	6.208	
	Node	1	2	3	10	11	12	
	Kinetic	10-4	10-4	10-4	10-6	10-6	10-6	
<i>x</i> ₂	Energy	1.016	1.076	1.016	6.278	7.929	6.278	
=	Helicity	10-5	10-11	10-5	10-5	10-11	10-5	
.23	CD 2 nd	-6.858	5.629	6.858	1.565	-2.675	-1.565	

CD 4 th	-4.628	5.552	4.628	1.477	-2.708	-1.477
Uni NL	-5.159	5.576	5.159	1.498	-2.701	-1.498
Uni L	-5.157	5.576	5.157	1.498	-2.701	-1.498
Enstrophy	10-3	10-3	10-3	10-4	10-5	10-4
CD 2 nd	6.174	1.360	6.174	2.269	8.395	2.269
CD 4 th	6.269	1.371	6.269	2.339	8.505	2.339
Uni NL	6.246	1.368	6.246	2.321	8.474	2.321
Uni L	6.245	1.368	6.245	2.321	8.474	2.321

Following the symmetry of the initial perturbation, equal values of kinetic energy and enstrophy are observed in planes $z_3 = 0.25$ and $z_3 = 0.75$, as well as opposite values of helicity, which vanishes at central plane $z_3 = 0.5$. Fluctuating variables generally decay from nodes at plane $x_1 = 1.5$ to corresponding nodes at plane $x_1 = 2.5$, but at level $x_2 = 0.75$, helicity and particularly enstrophy increase. At level $x_2 = 0.25$, helicity changes its sign between the two planes. Next Tables refer to node 4.

Table 2 refers to the kinetic energy equation. Advective and pressure term are expressed in advective forms Aa= $U_j \partial \overline{(u_i u_i/2)} / \partial x_j$, Ta= $\overline{u_j \partial (u_i u_i/2)} / \partial x_j$ and Pa= $\overline{u_j \partial p / \partial x_j}$ and in divergence forms Ad= $\partial (U_j \overline{u_i u_i/2}) / \partial x_j$, Td= $\partial \overline{(u_j u_i u_i/2)} / \partial x_j$ and Pa= $\partial \overline{u_j p} / \partial x_j$.

Each form of kinetic energy equation derives from one side of equality Vu=Vs, where $Vu = v\overline{u_i \partial^2 u_i / \partial x_j \partial x_j}$ and $Vs = 2v\overline{u_i \partial s_{ij} / \partial x_j}$. The first form is decomposed in identity Vu=Df-Du, where $Df = v \partial^2 \overline{u_i u_i / 2} / \partial x_j \partial x_j$ and $Du = v(\overline{\partial u_i / \partial x_j})(\partial u_i / \partial x_j)$; the second form according to Vs=Ts-Ds (Eq. 13), where $Ts = v \partial \overline{u_i s_{ij}} / \partial x_j$ and $Ds = v\overline{s_{ij} s_{ij}}$. The production term is $P = \overline{u_i u_j} S_{ij}$.

Differences such as $Aa \neq Ad$, $Vu \neq Df$ -Du etc., and the spreading of results among the schemes are put as percentage of the term with greater modulus. The divergence form turbulent transport (Td) according to generalized UNIFAES is computed as the difference between the mean advective terms according to non-linear and linear versions. The residuals of the transport equations are presented in absolute value and as percentage of the term with greater modulus.

Tomm		40x40x1	160		60x60x240				
$(\text{or } \neq)$	CD 2 nd	CD 4 th	Uni	Spread	CD 2 nd	CD 4 th	Uni	Spread	
(01 +)	ord.	ord.	L	ing	ord.	ord.	L	ing	
Aa (E-4)	-4.911	-4.910	-4.910	0.0	-4.986	-4.985	-4.986	0.0	
Ad (E-4)	-4.914	-4.911	-4.911	0.1	-4.987	-4.986	-4.985	0.0	
Aa≠Ad	0.1%	0.0%	0.0%		0.0%	0.0%	0.0%		
Ta (E-5)	1.788	1.837	1.835	5.7	1.891	1.914	1.911	1.7	
Td (E-5)	1.724	1.807	1.852	6.9	1.863	1.905	1.907	2.3	
Ta≠Td	3.6%	1.3%	0.9%		1.5%	0.5%	0.2%		
Pa (E-4)	-7.132	-7.148	-7.132	0.2	-6.938	-6.945	-6.938	0.1	
Pd (E-4)	-6.850	-7.056	-6.981	2.9	-6.816	-6.909	-6.875	1.3	
Pa≠Pd	4.0%	1.3%	2.1%		1.8%	0.5%	0.9%		
Df (E-4)	-1.328	-1.336	-1.328	0.9	-1.310	-1.314	-1.310	0.4	
Du (E-4)	4.127	4.248	4.228	2.8	4.073	4.126	4.116	1.3	
Vu (E-4)	-5.540	-5.588	-5.557	0.9	-5.420	-5.441	-5.430	0.4	
Vu≠Df-Du	1.5%	0.1%	0.0%		0.7%	0.0%	0.1%		
Ts (E-4)	-3.692	-3.826	-3.795	3.5	-3.767	-3.828	-3.814	1.6	
Ds (E-4)	1.508	1.573	1.561	4.1	1.485	1.513	1.508	1.9	
Vs (E-4)	-5.372	-5.573	-5.529	3.6	-5.374	-5.436	-5.416	1.1	
Vs≠Ts-Ds	3.2%	3.1%	3.1%		2.3%	1.7%	1.7%		
Vs≠Vu	3.0%	0.3%	0.5%		0.8%	0.1%	0.3%		
Pr (E-4)	6.817	6.837	6.832	0.3	6.629	6.637	6.635	0.1	
Resid. Eq.	-4.085	5.465	5.294		2.773	3.379	3.291		
12 (E-5)	5.7%	7.6%	7.4%		4.0%	4.9%	4.7%		
Resid. Eq.	-1.528	3.570	3.293		1.455	2.380	2.239		
13 (E-5)	2.1%	5.0%	4.6%		2.1%	3.4%	3.2%		

Table 2 – Statistics of kinetic energy transport equations for Re=600 at node 4 with different numerical schemes and algebraic forms.

Table 3 refers to the helicity equation. Advection and pressure terms of the equation admit advective and divergence forms: Aa= $U_j \partial \overline{u_i w_i} / \partial x_j$, Ad= $\partial (U_j \overline{u_i w_i}) / \partial x_j$, Ta= $\overline{u_j \partial (u_i w_i) / \partial x_j}$, Td= $\partial \overline{u_i w_i u_j} / \partial x_j$, Pa= $\overline{w_j \partial p / \partial x_j}$ and Pd= $\overline{\partial (pw_j) / \partial x_j}$. There are two production terms by interaction with mean velocity fields: P1= $\partial (u_i u_j) / \partial x_j W_i$ and P2 = $\overline{u_i u_j} \partial W_i / \partial x_j$. The fluctuating production term admits distinct algebraic forms: Pt1= $\overline{u_i w_j s_{ij}}$, Pt2= $\overline{w_j \partial (u_i u_i / 2) / \partial x_j}$ and Pt3= $\partial (\overline{w_j u_i u_i} / 2) / \partial x_j$. The viscous terms of the helicity equation used equality V=Df-Ds, with V= $v(\overline{u_i \partial^2 w_i / \partial x_i \partial x_i} + \overline{w_i \partial^2 u_i / \partial x_i \partial x_j})$, Df= $v \partial^2 \overline{u_i w_i} / \partial x_j \partial x_j$ and Ds = $v(\overline{\partial u_i / \partial x_j}) (\partial w_i / \partial x_j)$.

Term		40x40x	160		60x60x240			
	CD	CD	Uni	Spread	CD	CD	Uni	Spread
(or ≠)	2^{nd} ord.	4 th ord.	L	ing	2^{nd} ord.	4 th ord.	L	ing
Aa (E-3)	6.999	6.954	6.986	0.6	7.292	7.273	7.286	0.3
Ad (E-3)	7.003	6.955	6.983	0.7	7.293	7.274	7.284	0.3
Aa≠Ad	0.1%	0.0%	0.0%		0.0%	0.0%	0.0%	
Ta (E-4)	-3.001	-3.295	-3.223	10.0	-3.547	-3.699	-3.663	4.1
Td (E-4)	-3.063	-3.317	-3.276	7.7	-3.574	-3.709	-3.664	3.6
Ta≠Td	1.3%	0.5%	0.8%		0.0%	0.1%	0.2%	
Pa (E-2)	1.445	1.470	1.461	1.7	1.429	1.440	1.436	0.8
Pd (E-2)	1.439	1.470	1.462	2.1	1.426	1.440	1.436	1.0
Pa≠Pd	0.4%	0.0%	0.1%		0.2%	0.0%	0.0%	
Df (E-3)	5.817	5.920	5.862	1.7	5.716	5.760	5.734	0.8
Du (E-2)	-1.024	-1.060	-1.050	3.4	-1.013	-1.028	-1.024	1.5
Vu (E-2)	1.615	1.652	1.635	2.2	1.588	1.605	1.597	1.1
Vu≠Df-Du	0.6%	0.0%	0.1%		0.2%	0.1%	0.0%	
P1 (E-3)	4.973	5.095	5.064	2.4	4.944	4.998	4.985	1.0
P2 (E-3)	-1.232	-1.267	-1.258	2.8	-1.228	-1.244	-1.240	1.3
Pt1 (E-5)	-4.744	-5.681	-5.211	19.5	-6.473	-6.937	-6.731	6.7
Pt2 (E-5)	-4.801	-5.701	-5.637	23.1	-6.514	-6.942	-6.885	7.0
Pt3 (E-5)	-4.765	-5.629	-5.535	15.3	-6.490	-6.926	-6.858	6.3
≠Pt123	1.2%	1.3%	7.6%		0.6%	0.2%	2.2%	
Resid. Eq.	-1.071	-1.493	-1.366		-0.730	-0.915	-0.857	
(E-3)	7.4%	10.2%	9.3%		5.1%	6.4%	6.0%	

Table 3 – Statistics of helicity transport equations for Re=600 at node 4 with different numerical schemes and algebraic forms.

Table 4 – Statistics of enstrophy transport equations for Re=600 at node 4 with different numerical schemes and algebraic forms.

Term		60x60x240						
	CD	CD	Uni	Spread	CD	CD	Uni	Spread
(or ≠)	2^{nd} ord.	4 th ord.	L	ing	2^{nd} ord.	4 th ord.	L	ing
Aa (E-2)	-8.528	-8.584	-8.600	0.8	-8.701	-8.728	-8.732	0.4
Ad (E-2)	-8.530	-8.585	-8.598	0.8	-8.702	-8.729	-8.732	0.3
Aa≠Ad	0.0%	0.0%	0.0%		0.0%	0.0%	0.0%	
Ta (E-3)	2.763	3.629	3.368	23.9	3.641	4.075	3.959	10.7
Td (E-3)	3.066	3.743	3.399	18.1	3.774	4.113	3.959	8.2
Ta≠Td	9.9%	3.0%	0.9%		3.5%	0.9%	0.0%	
Df (E-2)	-4.918	-5.133	-5.027	4.2	-4.855	-4.948	-4.901	1.9
Du (E-2)	8.571	9.332	9.183	8.2	8.648	8.987	8.917	3.8
Vu (E-2)	-13.791	-14.484	-14.234	4.8	-13.637	-13.938	-13.833	2.2
Vu≠Df-Du	2.2%	0.1%	0.2%		1.0%	0.0%	0.1%	
P1 (E-2)	1.693	1.780	1.757	4.9	1.692	1.731	1.721	2.3
P2 (E-2)	8.002	8.175	8.149	2.1	7.912	7.987	7.974	0.9
P3 (E-2)	-1.345	-1.401	-1.383	4.0	-1.347	-1.372	-1.364	1.8
PT (E-4)	-8.787	-8.320	-8.757	7.9	-6.874	-6.607	-6.794	3.9

Resid. Eq.	1.861	11.619	8.521	2.255	6.603	5.250	
(E-3)	2.2%	12.5%	9.3%	2.6%	7.3%	5.9%	

Table 4 refers to the enstrophy equation. Mean and turbulent transport terms admit advective and divergent forms: Aa = $\partial (U_j \overline{w_i w_i/2}) / \partial x_j$, Ad= $U_j \partial \overline{w_i w_i/2} / \partial x_j$, Ta= $\overline{u_j \partial (w_i w_i/2)} / \partial x_j$ and Td = $\partial \overline{u_j w_i w_i/2} / \partial x_j$. The viscous terms results from identity V=Df-Ds, where V= $vw_i \partial^2 w_i / \partial x_j \partial x_j$, Df = $v \partial^2 (\overline{w_i w_i/2}) / \partial x_j \partial x_j$ and Ds= $v \overline{(\partial w_i/\partial x_j)} (\partial w_i/\partial x_j)$. There are three production terms due to interaction of mean and fluctuating field, P1= $\overline{w_i u_i} \partial W_i / \partial x_i$, P2= $\overline{w_i w_i S_{ij}}$ and P3= $\overline{w_i S_{ij}} W_i$, and one production term due to the fluctuating field only, Pt = $\overline{w_i w_i S_{ij}}$.

5. CONCLUSION

For all scalars, differences between advective and divergence forms of the advective term are negligible, as well as differences between the advective terms of the various schemes.

The differences between algebraic forms and the spreading of results between the measuring schemes reduce at ratio generally close to 9:4 from mesh 160x40x40 to mesh 240x60x60, as expected for quadratic behavior. The residuals of the transport equations appear to reduce in sub-quadratic fashion. Second order central differencing usually presented the greater discrepancies between algebraic forms, forth order central differencing the smallest, and UNIFAES intermediate, tending to be closer to the fourth order scheme. However, the second order central differencing was the scheme that presented the smallest unbalances, whilst the fourth order scheme presented the greatest.

This paradox derives from the fact that the various schemes are operating upon the field produced by the solver, which evolves with refinement. Differences between algebraic forms depend solely on the scheme, and in this matter the fourth order scheme shows its expected superiority. But any comparison involving varying refinement depends also on the solver.

Since the generalized UNIFAES scheme employs the interpolating curve of the solver, it may be considered as reference for the solver convergence errors. In this interpretation, the errors of second order central differencing upon the velocity field partially compensates the errors of solver itself, whilst the errors of forth order scheme added to the solver error, in node 4.

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7. RESPONSIBILITY NOTICE

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