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## **DETERMINATION OF GLOBAL EFFICIENCY WITHOUT/WITH SUPPLEMENTARY BURNING OF A THERMOELECTRIC PLANT WITH COMBINED CYCLE OF NATURAL GAS**

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**Abstract.** *The present work applies the energy analysis with the objective of determining the overall efficiency of the combined cycle thermoelectric plant installed in Panaíba, in the state of Piauí; as well as to determine the benefits and losses of energy by the different equipment that compose the system, which will be represented by a Sankey diagram of the systems without/with supplementary burning. Correction factors for shaft power, exhaust gas flow, relative humidity, among others, will be used in order to adapt the conditions to the operating location of the thermoelectric plant. The work was developed to steady-state operation and under time averaged (yearly) conditions. It is seen that the energy analysis shows the advantages of the use of natural gas in a combined cycle, which presents a better efficiency of 50,81 % when the thermoelectric plant is without supplementary burning, compared to 47,39 % when the plant thermoelectric is with supplementary burning.*

**Keywords:** *combined cycle, global efficiency, natural gas, supplementary burning, thermoelectric plant.*

### **1. INTRODUCTION**

The liberation of the electric market coupled with the growing commitment to the environment has driven the development of new generation techniques that involve lower investment costs and at the same time help reduce greenhouse gas emissions.

Currently, the need to reduce pollution has led to a strong commitment to the implementation of renewable energy as the main source of supply. These are a clean source of energy, but not enough to meet the growing energy demand.

The storage of energy in large quantities is not currently possible, for this reason, the electrical system is a dynamic system. A system in which it is necessary to produce a constant balance between production and demand. Therefore, due to the variability of renewable energies, it is necessary to have another form of energy production that ensures that we can satisfy the growing demand at all times.

Combined cycle power generation technology is one of the most efficient, with lower environmental impact and greater responsiveness. Such plants are very fast in terms of power refers to other conventional technologies (coal or nuclear), which are often used to meet demand or power failures. They are therefore the best option to meet renewable production plants, through which they may not always have all the power required. Thanks to its characteristics seems to be the best alternative to meet the demand with the least possible environmental impact (Antunes, 1999; Pantalena, 1997; Villela *et al.*, 2007b).

Thermoelectric plants, when operating on natural gas and emitting flue gases, are also causing environmental problems on a smaller scale compared to other fuels. Its components resulting from combustion harm human life, animals and plants. The main components generated from the combustion of these fuels are carbon oxides (CO and CO<sub>2</sub>), sulfur oxides (SO<sub>2</sub> and SO<sub>3</sub>) and nitrous oxides (NO and NO<sub>2</sub>, normally denoted by NO<sub>x</sub>). There is also the particulate matter (PM) that changes the temperature of the system and influences the growth of the plants. CO<sub>2</sub> and NO<sub>x</sub> emissions are directly related to the greenhouse effect, since the negative effects of SO<sub>2</sub> can be seen in the formation of acid rain. There are also other harmful components present in the combustion of some fuels, such as heavy metals, dioxins, etc., which in small concentrations are harmful to the environment (Villela, *et al.*, 2007a).

The waste from thermoelectric plants with gas turbines associated with condensing steam turbines has a great environmental impact, since they use fuels such as: coal, oil and natural gas. In the analysis, emissions of CO<sub>2</sub>, SO<sub>2</sub> and NO<sub>x</sub> were considered, and a comparison was made of the concentration of each of these products with the standards in Romania (Cardu *et al.*, 1999).

In the broadest sense, a combined cycle plant consists of the integration of two or more thermodynamic energy cycles, in order to obtain a more complete and efficient conversion of the supplied energy into work or power. The exhaust gas heat from the gas turbine is used to produce steam in the recovery boiler to power the steam turbine. Currently, advanced combined cycle plants employing 2 or 3 pressure levels for steam and with a temperature in the 420 to 650 °C range for exhaust gases can increase their thermal efficiency by up to 58 % (Franco *et al.*, 2002; Villela *et al.*, 2007b).

## 2. METHODOLOGY

### 2.1. Parameter corrections for selected gas turbines

The parameters of the selected gas turbines must be corrected according to the local conditions of the plant installation, because the ambient temperature, the altitude and the relative humidity influence the performance of the turbine. Thus, the following conditions were adopted (Antunes, 1999; Pantalena, 1997):

Ambient temperature of 27 °C, altitude of 900 m (local conditions of the plant), and relative humidity 60 %.

According to (Brooks, 1994; Antunes, 1999), the correction factors are used according to the following equations shown in Tab. 1:

Table 1. Parameter corrections for selected gas turbines

| Parameter  | Correction equation  |
|--|--|
| From the ambient temperature on the shaft power        | $fc_{SP}^T = -0,004 * T_{amb} + 1,06$ (1)                                    |
| From ambient temperature to exhaust gas mass flow rate | $fc_{MF}^T = -0,006 * T_{amb} + 1,07$ (2)                                    |
| From the ambient temperature on the Heat Rate          | $fc_{HR}^T = 0,002 * T_{amb} + 0,97$ (3)                                     |
| From the altitude (A) on the shaft power               | (200 m ≤ A ≤ 2800 m)<br>$fc_{SP}^A = -0,0001 * A + 0,9816$ (4)               |
|  | (A < 200 m ou 2800 m ≤ A ≤ 3000 m)<br>$fc_{SP}^A = -0,0001 * A + 0,9975$ (5) |

### 2.2. Determination of turbine parameters for ISO and corrected conditions

ISO - Standard pressure and temperature condition. The equations are shown in Tab. 2 (Pantalena, 1997):

Table 2. Turbine parameters for ISO and corrected conditions

| Parameter  | Correction equation  |
|--|--|
| Gas turbine efficiency   | $n_T(\%) = \left( \frac{3600}{HEAT\ RATE} \right) * 100\%$ (6)   |
| Mass fuel flow   | $m_F \left( \frac{kg}{s} \right) = \left( \frac{HEAT\ RATE * E_P}{n_{generator} * PCI * 3600} \right), n_{generator} = 0,95$ (7) |
| Total heat of gases  | $Q_G = m_G * Cp_G * (T_4 - T_G), T_G = 150 + T_{PP}$ (8)   |
| Total heat of steam  | $Q_S = m_S * (h_6 - h_9), 16 \frac{kg}{s} * (3454 - 219,6) \frac{kJ}{kg} = 51750,4 \frac{kJ}{s}$ (9)                             |
| Supplemental fuel  | $m_{SF} = \frac{Q_S - Q_G}{n_{RB} * PCI}, n_{RB} = 0,80$ (10)  |
| Corrected parameters   | $E_P^* = E_P * FC_4 * FC_1$ (11)   |
|  | $m_G^* = m_G * FC_2$ (12)  |
|  | $m_F^* = m_F * FC_4$ (13)  |
|  | $T_4^* = T_4 * FC_2$ (14)  |
| Corrected Heat Rate  | $HEAT\ RATE^* \left( \frac{kJ}{kWh} \right) = \frac{PCI * m_F^* * 3600}{E_P^*}$ (15)   |
| Corrected gas turbine efficiency   | $n_T^*(\%) = \left( \frac{3600}{HEAT\ RATE^*} \right) * 100\%$ (16)  |
| Corrected total gas heat   | $Q_G^* = m_G^* * Cp_{GE}^* * (T_4^* - T_G)$ (17)   |
| Corrected supplemental fuel  | $m_{SF}^* = \frac{Q_S - Q_G^*}{n_{RB} * LHV}$ (18)   |
| Gas mass flow rate as a function of the gas temperature at the outlet of the gas turbine (Antunes, 1999) | $\dot{m}_G = \frac{\dot{m}_S * (h_6 - h_9)}{n_{RB} * Cp_G * (T_4 - T_G)}$ (19)   |

The formulas presented will be used for the development of each turbine selected from the (Gas Turbine World Handbook, 2012) described in references as (Biasi, 2012).

### 2.3. First law analysis of thermodynamics

The proposed systems operate on a permanent basis; all or adiabatic components, operate without loss of heat. Two decision variables are used: temperature ( $T_4$ ) of the exhaust gases in the gas turbine and the pressure ratio of the selected turbine ( $PR = P_3/P_4$ ). These variables were chosen by the thermodynamic influence of the system and also according to the criteria used in the selection of the gas turbine system. Thus,  $P_1$  and  $P_5$  equal to ambient pressure 0,101325 MPa, and  $\Delta P_{RB}$  and  $\Delta P_{CC}$  are the gas pressure losses in the recovery boiler and in the combustion chamber respectively (Franco *et al.*, 2002; Silveira *et al.*, 2003; Villela *et al.*, 2007a, 2007b). The equations are presented in Tab. 3:

Table 3. Equations of the analysis of the first law of thermodynamics

| Parameter   | Equation   |
|---|--|
| Turbine input temperature   | $T_3 = \frac{T_4}{\left\{ 1 - n_{ISO_{GT}} \left[ 1 - (PR)^{\frac{(1-\lambda_G)}{\lambda_G}} \right] \right\}} \quad (20)$   |
| Ratio of specific heats   | $\lambda_G(T) = \frac{1}{1 - \frac{R_G}{C_{P_G}(T)}}, \lambda_G \left( \frac{C_{P_G}}{C_{V_G}} \right) \quad (21)$   |
| Turbine output temperature  | $T_2 = T_1 * \left\{ 1 + \frac{1}{n_{ISO_{GT}}} \left[ \left( \frac{P_2}{P_1} \right)^{\frac{(\lambda_{air}-1)}{\lambda_{air}}} - 1 \right] \right\} \lambda_{air}(T) = \frac{1}{1 - \frac{R_{air}}{C_{P_{air}}(T)}} \quad (22)$ |
| Ratio of specific heats of the air  | $\lambda_{air}(T) = \frac{1}{1 - \frac{R_{air}}{C_{P_{air}}(T)}} \quad (23)$   |
| The specific enthalpies of air and gases in points 2, 3   | $h_2 = C_{P_{air}}(T_2) * (T_2 - T_0) + h_0 \quad (24)$  |
|   | $h_3 = C_{P_G}(T_3) * (T_3 - T_0) + h_0 \quad (25)$  |
| Isentropic process at the outlet of the steam turbine, with $P_7$ and enthalpies of liquid and vapor. | $s_7 = (1 - x_7) * sl_7 + x_7 * ss_7 \quad (26)$   |
|   | $h_{7\ ideal} = (1 - x_7) * hl + x_7 * hs_7 \quad (27)$  |
|   | $n_{ST} = \frac{(h_6 - h_{7\ real})}{(h_6 - h_{7\ ideal})} \quad (28)$   |
| Air mass flow   | $m_{air} = m_G - m_f \quad (29)$   |
| Pump and compressor work  | $W_P = \frac{m_s * (h_9 - h_8)}{n_p} \quad (30)$   |
|   | $W_C = m_{air} \int_{T_1}^{T_2} C_{P_{air}}(T) dt \quad (31)$  |
| The shaft power ( $W_{ST}$ ), electricity ( $Ep_{ST}$ ) and $n_{ST}$ is 95 %                          | $W_{ST} = m_s * (h_6 - h_7) \quad (32)$  |
|   | $Ep_{ST} = W_{ST} * n_{ST} \quad (33)$   |
| Power in the gas turbine ( $Ep_{GT}$ ) with a $n_{GT}$ of 98 %  | $Ep_{GT} = W_{GT} * n_{GT} \quad (34)$   |
|   | $Ep_{TOTAL} = Ep_{GT} + Ep_{ST} \quad (35)$  |
| Power supplied by the fuel  | $E_f = m_f * LHV \quad (36)$   |
|   | $E_f = E_f + E_{sup\ fuel} \quad (37)$   |
| Overall efficiency of plant without/with supplementary burning  | $n_{OP} = \frac{Ep_{GT} + Ep_{ST} - W_P}{E_{fuel}} \quad (38)$   |
|   | $n_{OP} = \frac{Ep_{GT} + Ep_{ST} - W_P}{E_{fuel} + E_{sup\ fuel}} \quad (39)$   |

## 3. RESULTS AND DISCUSSIONS

### 3.1. Description of thermoelectric plant

The scheme of a gas turbine combined cycle plant associated with the recovery boiler without supplementary burning and steam turbine is shown in Fig. 1.

The proposed scheme presents seven components of the thermal plant: compressor, combustion chamber, gas turbine, recovery boiler, steam turbine, condenser and pump. In atmospheric conditions the air enters the compressor and is compressed to the combustion pressure. Subsequently, it is sent to the combustion chamber where the fuel is continuously burned under constant pressure and the combustion gases expand in the gas turbine producing electricity. Exhaust gases are then directed to the recovery boiler, where superheated steam is produced and sent to the steam turbine, producing even more electrical energy (Villela *et al.*, 2007a, 2007b).

Figure 2 shows the gas turbine system associated with the recovery boiler with supplementary burning and steam turbine. It is verified that the process is identical to that of Fig. 1, but in the recovery boiler a value of 30 % of the fuel (natural gas) consumed in the combustion chamber of the gas turbine system is set for the additional firing.

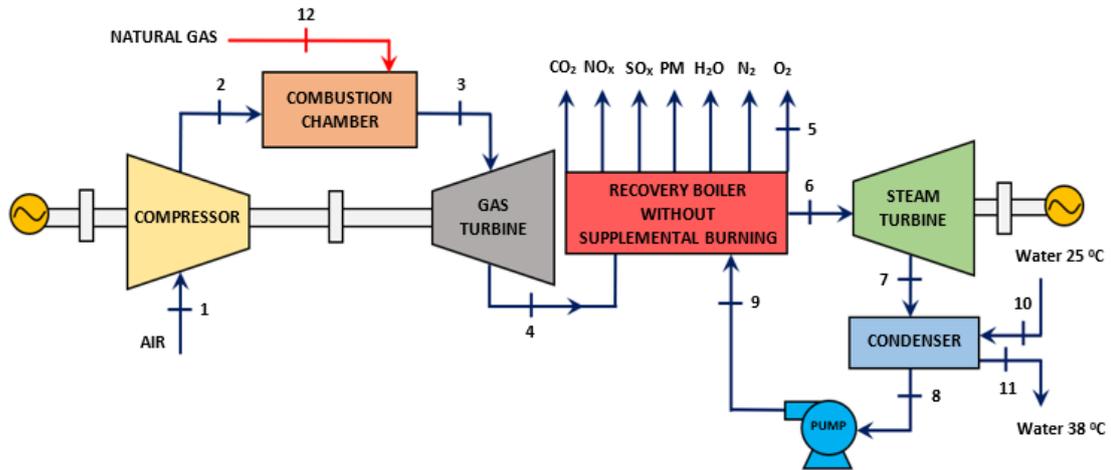


Figure 1. Gas turbine system associated with the recovery boiler without supplementary burning and steam turbine.

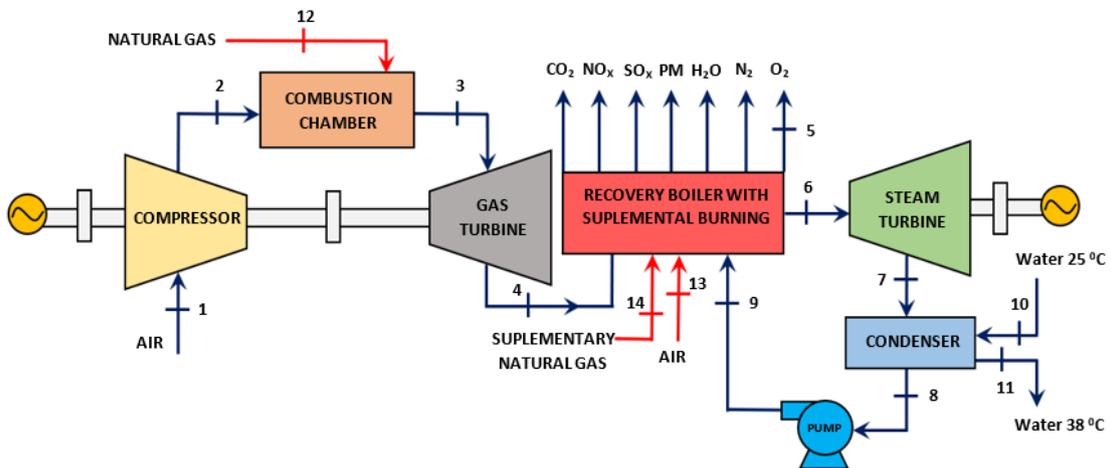


Figure 2. Gas turbine system associated with the recovery boiler with supplementary burning and steam turbine.

### 3.2. Selection of steam turbine

Among the parameters necessary for the comparison of the thermoelectric plants, using natural gas, are important mainly those related to the selected turbine. The choice of steam turbine used was based on the manufacturer's catalog (Siemens, 2018) on the market. Table 4 shows the selected steam turbine. The machine was chosen in view of the three temperature and pressure levels of the steam at the steam turbine inlet.

Table 4. Selected steam turbine (Siemens, 2018)

| Parameters       | Values  |
|------------------|---------|
| Temperature (°C) | 540     |
| Pressure (MPa)   | 12      |
| Velocity (rpm)   | 14600   |
| Model            | SST-200 |
| Manufacturer     | Siemens |
| Power (kW)       | 20000   |

### 3.3. Selection of gas turbine

In this section, the gas turbine is selected according to the parameters of Tab. 5 and then we continue with the procedure.

Table 5. Input data for the selection of the gas turbine

| Physical parameters  | Symbology  | Specification  | Units                       |
|--|------------|----------------|-----------------------------|
| Exhaust gas temperature with subsequent correction by the Pinch Point method.  | $T_G$      | 150            | °C                          |
| Enthalpy at point 6 superheated steam determined as a function of the temperature and inlet pressure of the steam turbine. | $h_6$      | -              | kJ/kg                       |
| Enthalpy at point 9 for a feed water temperature entering the boiler at 50 °C and pressure of 12 MPa, compressed liquid.   | $h_9$      | -              | kJ/kg                       |
| Enthalpy at point 8 for a temperature of 29,8 °C and pressure of 0,005 MPa: compressed liquid.                             | $h_8$      | -              | kJ/kg                       |
| Pressure in point 7 (Siemens, 2018)  | $P_7$      | 0,005          | MPa                         |
| Recovery boiler efficiency without supplementary burning (Silveira, 2003).   | $H_{RC}$   | 70             | %                           |
| Recovery boiler efficiency without supplementary burning (Silveira, 2003).   | $H_{RC}$   | 90             | %                           |
| Steam turbine steam flow (project data)  | $m_{ST}$   | 16             | kg/s                        |
| Specific heat exhaust gas  | $C_{pG}$   | 1,14           | kJ/kg K                     |
| Specific heat at constant pressure of natural gas  | $C_{pNG}$  | 1,209          | kJ/kg K                     |
| Lower Heating Value  | $LHV_{NG}$ | 47966<br>37000 | kJ/kg<br>kJ/Nm <sup>3</sup> |

Table 6 it shows the enthalpy values to determine the solution of Eq. (19), and to obtain the  $T_4$  as a function of the mass flow of the exhaust gases.

Table 6. Pressure enthalpies and operating temperatures

| P (MPa) | T (°C) | h (kJ/kg)     |
|---------|--------|---------------|
| 12      | 540    | $h_6 = 3454$  |
| 0,005   | X=0,9  | $h_7 = 2319$  |
| 0,005   | X=0,0  | $h_8 = 137,8$ |
| 12      | 50     | $h_9 = 219,6$ |

$$\dot{m}_G = \frac{16 \frac{kg}{s} * (3454 - 219,6) \frac{kJ}{kg}}{0,9 * 1,14 \frac{kJ}{kgK} * (T_4 - 150) ^\circ C} \quad (40)$$

$$\dot{m}_G = \frac{50438,9 \text{ kg}}{(T_4 - 150) \text{ s}} \quad (41)$$

According to Eq. (41), the best gas turbine is determined by varying the temperature  $T_4$  in a range of 400 to 655 °C. The data used are ISO ( $T = 15$  ° C, altitude = 0 m and 60 % relative humidity) and are available in the “Gas Turbine Handbook” (Biasi, 2012). The results are shown in Tab. 7.

Table 7. Mass gas flow in function of the maximum temperature in the gas turbine

| $T_4$ (°C) | $\dot{m}_G$ (kg/s) |
|------------|--------------------|
| 400        | 201,8              |
| 450        | 168,1              |
| 500        | 144,1              |
| 550        | 126,1              |
| 600        | 112,1              |
| 650        | 100,9              |

Table 8, shows the gas turbine systems selected from the “Gas Turbine World Handbook” (Biasi, 2012)., under the ISO conditions.

The tables and figures below show the gas flow ( $\dot{m}_G$ ) as a function of the temperature at the outlet of the selected turbines, considering the natural gas.

Table 8. Gas turbine systems selected from Gas Turbine World Handbook, 2012

| GT | Model   | Year | $W_{liq-Ep}$<br>(kW) | HR<br>(kJ/kWh) | $m_G$<br>(kg/s) | $T_G$<br>(°C) | $P_r$ |
|----|---|------|----------------------|----------------|-----------------|---------------|-------|
| A  | GE Energy Oil & Gas: LM6000PG                     | 2008 | 51204                | 8590           | 143,9           | 470           | 30    |
| B  | IHI Power Systems: LM6000PH Sprint                | 2010 | 51342                | 8774           | 139,9           | 470           | 35    |
| C  | Pratt & Whitney Power Systems:<br>SWIFTPAC 50 DLN | 2003 | 51235                | 9395           | 169,6           | 458           | 19,5  |
| D  | Rolls-Royce: Trent 60 DLE                         | 1996 | 51685                | 8586           | 154,6           | 440           | 34    |
| E  | Siemens Energy: SGT-900                           | 1982 | 49500                | 11025          | 175,1           | 514           | 15,3  |

The values shown will be plotted on the thermal parity curve of Fig. 3.

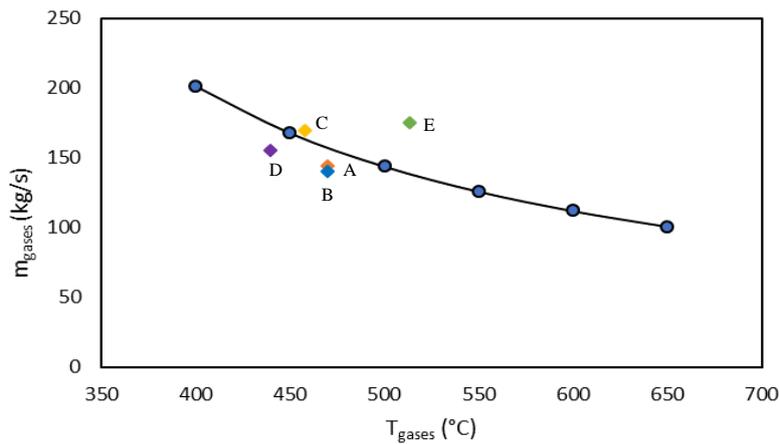


Figure. 3. Thermal parity curve with turbines selected from the Gas Turbine World Handbook, 2012.

### 3.4. Correction Factors for selected gas turbine parameters

The combined cycle thermoelectric plant will have an installed capacity of 60 MW and will be installed in Panaíba, in the state of Piauí. Which has the following climatic conditions: Ambient temperature 27 °C, 1 atmosphere pressure, 900 m altitude and 60 % relative humidity. The work was developed to steady-state operation and under time averaged (yearly) conditions. Using the equations of Tab. 1. The values of the correction factors of the local conditions (ambient temperature, altitude and relative humidity) of the plant installation are presented in Tab. 9:

Table 9. Results of correction factors for local conditions

| CF              | Correction Factors                                     | Values |
|-----------------|--|--------|
| CF <sub>1</sub> | From the ambient temperature on the power axis         | 0,952  |
| CF <sub>2</sub> | From ambient temperature to exhaust gas mass flow rate | 0,908  |
| CF <sub>3</sub> | From the ambient temperature on the Heat Rate          | 1,024  |
| CF <sub>4</sub> | From the altitude (A) on the shaft power               | 0,892  |

We apply the values of the correction factors to the gas turbines selected from the Gas Turbine World Handbook (Biasi, 2012), shown in the Tab. 10.

Table 10. Gas Turbine Corrected Values Selected from the Gas Turbine World Handbook, 2012

| GT | Model                                | Year | $W_{liq-Ep}^*$<br>(kW)<br>CF <sub>1</sub> e CF <sub>4</sub> | HR*<br>(kJ/kWh)<br>CF <sub>3</sub> | $m_G^*$<br>(kg/s)<br>CF <sub>2</sub> | $T_G^*$<br>(°C)<br>CF <sub>2</sub> | $P_r$ |
|----|--------------------------------------|------|---|------------------------------------|--------------------------------------|------------------------------------|-------|
| A  | GE Energy Oil & Gas: LM6000PG        | 2008 | 43482   | 8796                               | 130,7                                | 427                                | 30    |
| B  | IHI Power Systems: LM6000PH Sprint   | 2010 | 43599   | 8985                               | 127,1                                | 427                                | 35    |
| C  | Pratt & Whitney Power Systems:50 DLN | 2003 | 43508   | 9620                               | 153,9                                | 416                                | 19,5  |
| D  | Rolls-Royce: Trent 60 DLE            | 1996 | 43890   | 8792                               | 140,4                                | 400                                | 34    |
| E  | Siemens Energy: SGT-900              | 1982 | 42035   | 11290                              | 158,9                                | 467                                | 15,3  |

The values shown will be plotted on the new thermal parity curve of Fig. 4.

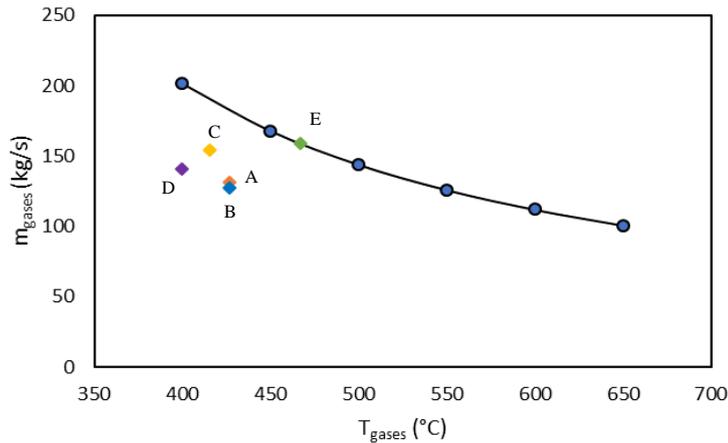


Figure 4. Thermal parity curve - corrected with turbines selected from the Gas Turbine World Handbook, 2012.

According to Fig. 4, the points below the curve are more likely to reach the thermal parity with the use of a supplementary burner and they are subjected to evaluation.

### 3.5. Determination of Pinch-Point

The temperature of the gases at the exit of the recovery boiler may not be less than a minimum value, which depends on the fuel and the combustion process, since acids may form when the products of combustion are cooled in the boiler, damaging the equipment due to the composition of the fuel (in the case of natural gas the formation of acid hardly exists). Therefore, the temperature ( $T_5$ ) of the exhaust gases in an already selected gas turbine model should be corrected by the *pinch-point* according to the following criteria, obtained from Fig. 5 (Balestieri, 1994; Barclay, 1995).

Figure 5 shows the gas cooling profile up to the intersection with the T axis denoted by  $T_{PP}$ . The temperature  $T_4$  is the already corrected gas temperature at the outlet of the gas turbine and  $T_5$  is the saturation temperature, under the saturation pressure condition.

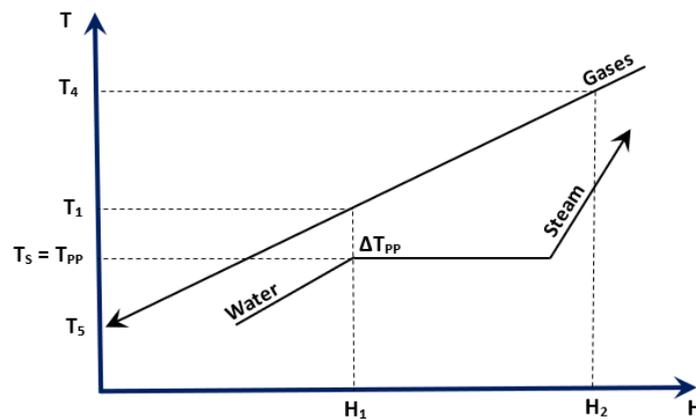


Figure 5. Determination of Pinch Point.

In order not to incur a thermodynamic impropriety, a minimum delta T ( $\Delta T_{PP}$ ) must be obtained for the gas cooling profile from the turbine. According to (Sue and Chuang, 2004) this value can be between 10 and 30 °C. A delta value of 25 °C was used for this delta, determining the dew point temperature ( $T_1$ ):

$$T_1 = T_5 + \Delta T_{PP} \quad (42)$$

The temperature of the boiler exhaust gas is determined by (Antunes, 1999; Barclay, 1995; Balestieri, 1994):

$$T_G = T_4 - \frac{T_4 - T_1}{H_2 - H_1} * H_2 \quad (43)$$

Where: the rates of heat transfer received ( $H_2$ ) and assigned ( $H_1$ ) are respectively given by:

$$H_2 = \dot{m}_G * C_{pG(T_4)} * T_4 \quad (44)$$

$$H_1 = \dot{m}_S * h_l \quad (45)$$

Where:  $h_l$  is the specific enthalpy of the saturated liquid, under the saturation pressure condition.

### 3.6. Turbine parameters for ISO and corrected conditions

The formulas presented in Tab. 2 will be used for the development of each turbine selected from the Gas Turbine World Handbook, 2012.

Table 11. Turbine data at ISO and corrected (\*) conditions for local parameters

| Turbines  | HR<br>(kJ/kWh) | $E_P$<br>(kW) | $n_T$<br>(%) | $T_4$<br>(°C) | $Q_G$<br>(kJ/s) | $m_G$<br>(kg/s) | $m_F$<br>(kg/s) | $m_{SF}$<br>(kg/s) | $m_{TF}$<br>(kg/s) |
|-----------|----------------|---------------|--------------|---------------|-----------------|-----------------|-----------------|--------------------|--------------------|
| A         | 8590           | 51204         | 41,9         | 470           | 48393,6         | 143,9           | 2,681           | 0,339              | 3,020              |
| A*        | 9495           | 43482         | 37,9         | 427           | 37547,5         | 130,7           | 2,391           | 0,566              | 2,957              |
| B         | 8774           | 51342         | 41,0         | 470           | 47048,4         | 139,9           | 2,746           | 0,218              | 2,964              |
| B*        | 9699           | 43599         | 37,1         | 427           | 36513,3         | 127,1           | 2,449           | 0,587              | 3,036              |
| C         | 9395           | 51235         | 38,3         | 458           | 54813,1         | 169,6           | 2,934           | 0,056              | 2,990              |
| C*        | 10387          | 43508         | 34,7         | 416           | 42282,5         | 153,9           | 2,617           | 0,467              | 3,084              |
| D         | 8586           | 51685         | 41,9         | 440           | 46704,7         | 154,6           | 2,705           | 0,225              | 2,930              |
| <u>D*</u> | <u>9494</u>    | <u>43890</u>  | <u>37,9</u>  | <u>400</u>    | <u>36012,6</u>  | <u>140,4</u>    | <u>2,413</u>    | <u>0,389</u>       | <u>2,802</u>       |
| E         | 11025          | 49500         | 32,7         | 514           | 67669,1         | 175,1           | 3,327           | 0,000              | 3,327              |
| E*        | 12192          | 42035         | 29,5         | 467           | 52894,6         | 158,9           | 2,968           | 0,246              | 3,214              |

From Tab. 11, the cycle with the turbine D spends less fuel (2,802 kg/s), has a better efficiency (37,9 %) and still meets the electric demand of 40000 kW.

### 3.7. Analyzing the first law of thermodynamics

Table 12 shows the fixed parameters for the following analysis.

Table 12. Fixed parameters for energy analysis

| Physical parameters                            | Symbology       | Specification | Units |
|--|-----------------|---------------|-------|
| Compressor efficiency                          | $n_C$           | 86            | %     |
| Pump Efficiency                                | $n_P$           | 90            | %     |
| Ratio of natural gas                           | $R_{NG}$        | 0,29          | -     |
| Ratio of air                                   | $R_{air}$       | 0,298         | -     |
| Lower calorific value of natural gas           | $LHV_{GN}$      | 47966         | kJ/kg |
| Loss of gas pressure in the recovery boiler    | $\Delta P_{RB}$ | 0,05          | -     |
| Loss of gas pressure in the combustion chamber | $\Delta P_{CC}$ | 0,05          | -     |
| Gas turbine ISO efficiency                     | $n_{ISO\ GT}$   | 89            | %     |
| Compressor ISO efficiency                      | $n_{ISO\ C}$    | 80            | %     |
| Steam turbine ISO efficiency                   | $n_{ISO\ ST}$   | 86            | %     |
| Recovery Boiler Efficiency                     | $n_{RB}$        | 80            | %     |
| Efficiency of the combustion chamber           | $n_{CC}$        | 99            | %     |
| Steam turbine efficiency                       | $n_{ST}$        | 98            | %     |
| Gas turbine efficiency                         | $n_{GT}$        | 87            | %     |

Table 13, shows the results of the energy analysis using the equations in Tab. 3 and the fixed parameters in Tab. 12, considering the system without/with supplementary burning, respectively.

Figures 6 and 7 show the Sankey diagrams with system efficiencies, both for a thermal cycle without supplementary and supplementary burning, respectively.

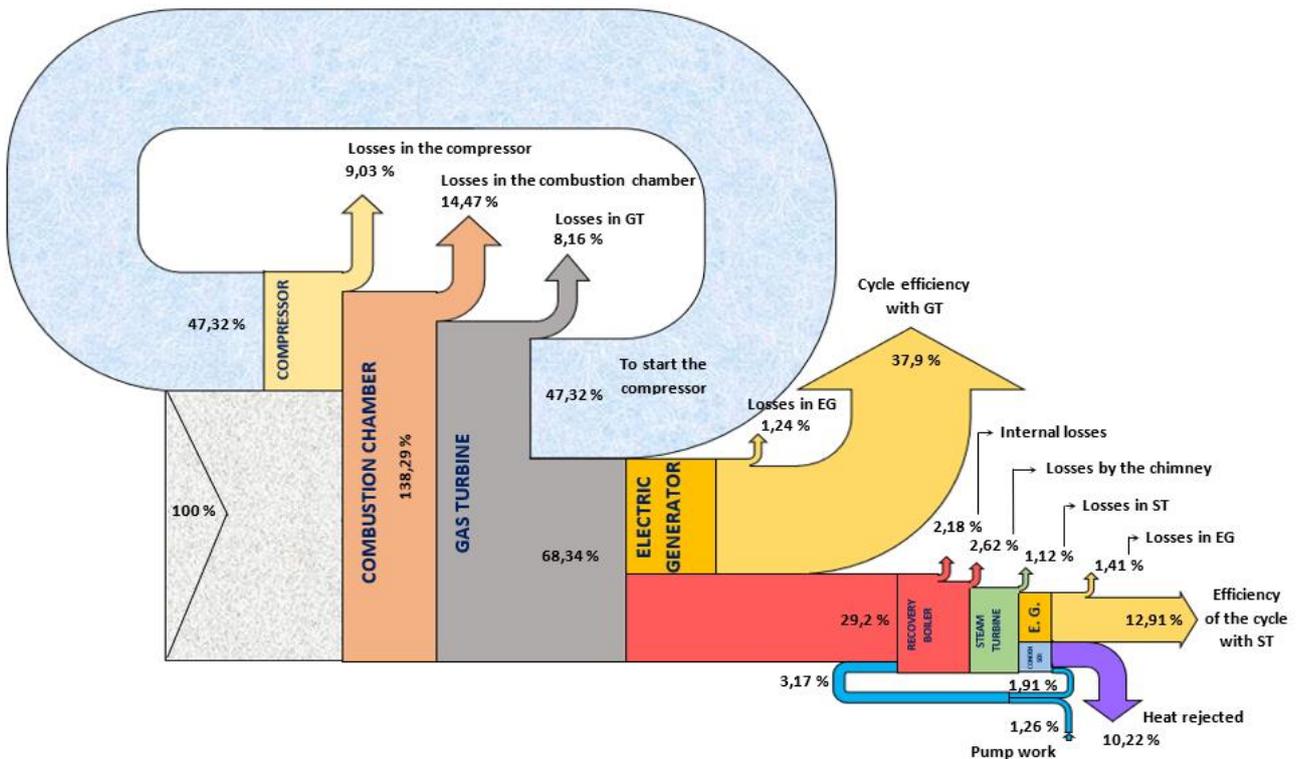


Figure 6. Sankey diagram of the combined cycle thermal plant without supplemental burning.

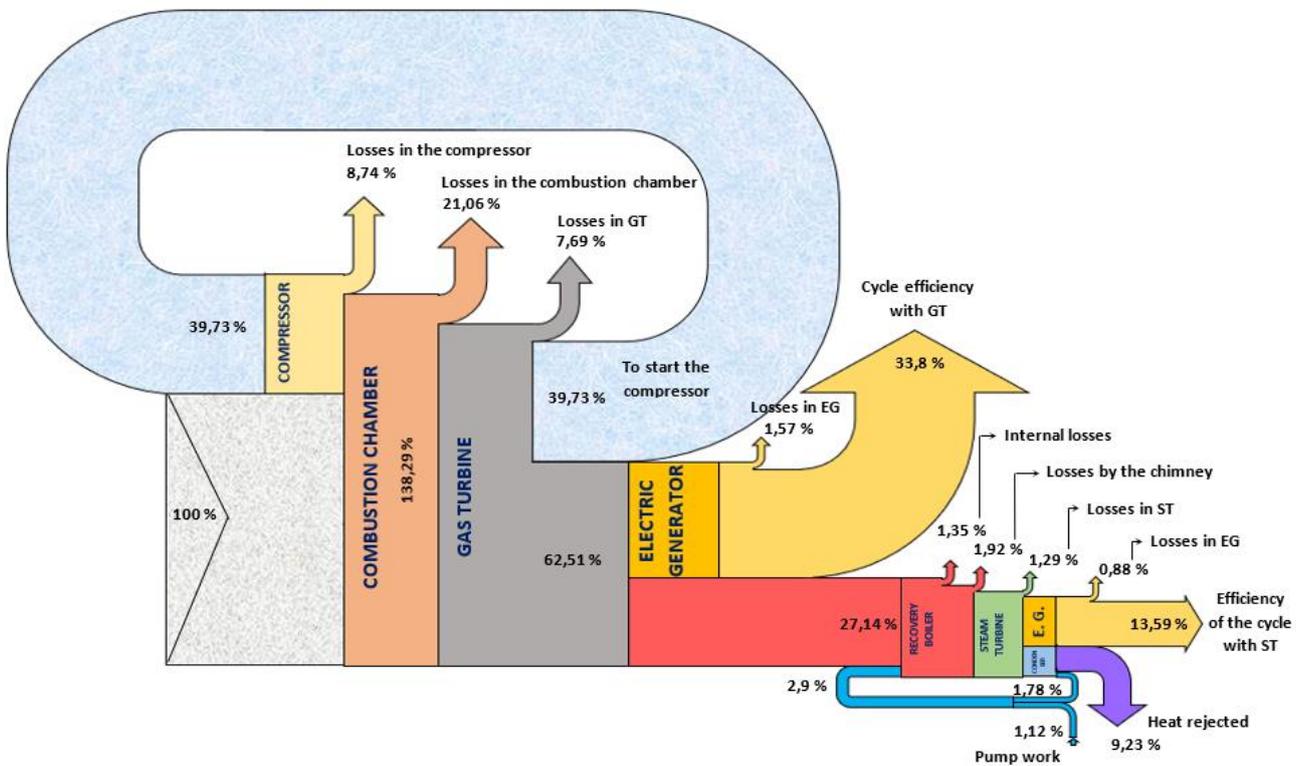


Figure 7. Sankey diagram of the combined cycle thermal plant with supplemental burning.

Table 13. Result of energy analysis the system without/with supplementary burning

| Energy analysis           | Without supplementary burning | With supplementary burning |
|---------------------------|-------------------------------|----------------------------|
| $E_{\text{fuel}}$ (kW)    | 115741,9                      | 134400,7                   |
| $E_{P\text{ GT}}$ (kW)    | 43012,2                       | 45451,1                    |
| $E_{P\text{ ST}}$ (kW)    | 17252,0                       | 19796,8                    |
| $W_P$ (kW)                | 1454,2                        | 1499,7                     |
| $E_{P\text{ TOTAL}}$ (kW) | 60264,2                       | 65247,9                    |
| $\eta_{\text{GL}}$ (%)    | 50,81                         | 47,39                      |

#### 4. CONCLUSIONS

As expected, through energy analysis it is possible to verify that the best overall efficiency is obtained when using natural gas in the case (Rolls-Royce gas turbine: Trent 60 DLE associated with the Siemens SST-200 steam turbine) presenting a higher efficiency (50,81 %) compared to the system with supplementary burning (47,39 %). The two systems were represented in a Sankey diagram.

For future work, a simulation of the process could be carried out, taking into account the time, the annual meteorological variations for an estimate of production and efficiency closer to each season of the year.

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#### 7. RESPONSIBILITY NOTICE

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